

Selected Topics in Numerical Linear Algebra and Control

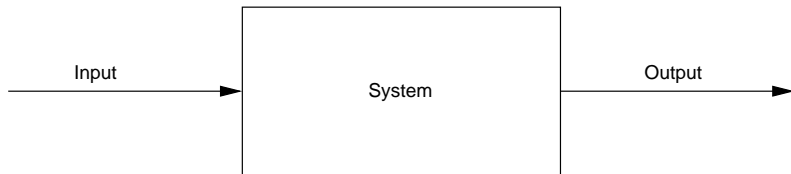
Web page: `http://web.math.hr/~kressner/nlacontrol`

Zlatko Drmač and Daniel Kressner
`{drmac, kressner}@math.hr`

The goal of this course is to

- ▶ cover all important aspects of linear control systems;
- ▶ emphasize numerical algorithms, in particular from numerical linear algebra;
- ▶ lead to state-of-the-art techniques, e.g., in model reduction and optimal control.

Systems are separated from their outer environment.



Input variables:

External quantities acting on the system.

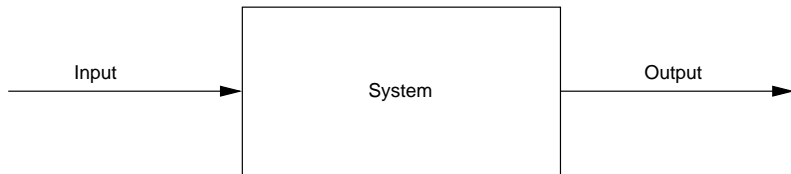
State variables:

Describe condition or state of the system.

Output variables:

System quantities that can be observed.

Systems are separated from their outer environment.



Input variables:

External quantities acting on the system.

State variables:

Describe condition or state of the system.

Output variables:

System quantities that can be observed.

Continuous time control systems

$$\begin{aligned}\dot{x}(t) &= f(t, x(t), u(t)), & x(0) &= x_0, \\ y(t) &= h(t, x(t), u(t)),\end{aligned}$$

where

$$\begin{aligned}t \in [0, \infty) &= \text{time}, \\ x(t) \in \mathbb{R}^n &= \text{state vector at time } t, \\ u(t) \in \mathbb{R}^m &= \text{input vector at time } t, \\ y(t) \in \mathbb{R}^p &= \text{output vector at time } t,\end{aligned}$$

and $f : \mathbb{R} \times \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^n$, $h : \mathbb{R} \times \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^p$.

System = state equation (parameterized first-order ODE)
+ output equation (function in time/state/input)

Continuous time control systems

$$\begin{aligned}\dot{x}(t) &= f(t, x(t), u(t)), & x(0) &= x_0, \\ y(t) &= h(t, x(t), u(t)),\end{aligned}$$

where

$$\begin{aligned}t \in [0, \infty) &= \text{time}, \\ x(t) \in \mathbb{R}^n &= \text{state vector at time } t, \\ u(t) \in \mathbb{R}^m &= \text{input vector at time } t, \\ y(t) \in \mathbb{R}^p &= \text{output vector at time } t,\end{aligned}$$

and $f : \mathbb{R} \times \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^n$, $h : \mathbb{R} \times \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^p$.

System = state equation (parameterized first-order ODE)
+ output equation (function in time/state/input)

Extensions

This is *not* the most general form of a continuous time control systems.

- ▶ **higher order systems** (state equation involves higher order derivatives of x);
- ▶ **differential-algebraic systems** (implicit state equation $F(t, x(t), \dot{x}(t), u(t)) = 0$);
- ▶ **infinite-dimensional systems** (state equation is PDE).

Extensions

This is *not* the most general form of a continuous time control systems.

- ▶ **higher order systems** (state equation involves higher order derivatives of x);
- ▶ **differential-algebraic systems** (implicit state equation $F(t, x(t), \dot{x}(t), u(t)) = 0$);
- ▶ **infinite-dimensional systems** (state equation is PDE).

Extensions

This is *not* the most general form of a continuous time control systems.

- ▶ **higher order systems** (state equation involves higher order derivatives of x);
- ▶ **differential-algebraic systems** (implicit state equation $F(t, x(t), \dot{x}(t), u(t)) = 0$);
- ▶ **infinite-dimensional systems** (state equation is PDE).

Linear time-invariant (LTI) continuous time control systems

$$\begin{aligned}\dot{x}(t) &= Ax(t) + Bu(t), & x(0) &= x_0, \\ y(t) &= Cx(t) + Du(t),\end{aligned}$$

where

$$\begin{aligned}A \in \mathbb{R}^{n \times n} &= \text{state matrix,} \\ B \in \mathbb{R}^{n \times m} &= \text{input matrix,} \\ C \in \mathbb{R}^{p \times m} &= \text{output matrix,} \\ D \in \mathbb{R}^{p \times m} &= \text{feedthrough matrix (often zero).}\end{aligned}$$

Linear time-invariant (LTI) continuous time control systems

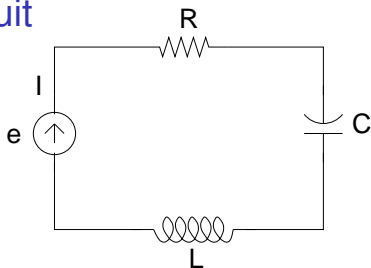
$$\begin{aligned}\dot{x}(t) &= Ax(t) + Bu(t), & x(0) &= x_0, \\ y(t) &= Cx(t) + Du(t),\end{aligned}$$

where

- $A \in \mathbb{R}^{n \times n}$ = state matrix,
- $B \in \mathbb{R}^{n \times m}$ = input matrix,
- $C \in \mathbb{R}^{p \times m}$ = output matrix,
- $D \in \mathbb{R}^{p \times m}$ = feedthrough matrix (often zero).

$$\Sigma = \left[\begin{array}{c|c} A & B \\ \hline C & D \end{array} \right] = \left[\begin{array}{|c|c|} \hline & \\ \hline & \\ \hline & \\ \hline & \\ \hline \end{array} \right]$$

Linear RLC circuit



Kirchhoff's law states that the total voltage must be balanced:

$$e(t) - V_R(t) - V_C(t) - V_L(t) = 0.$$

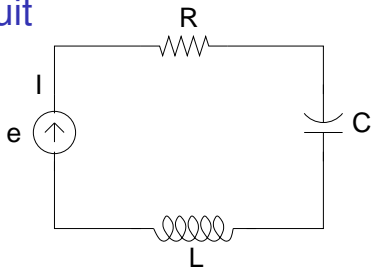
Combined with

$$V_R(t) = I(t)R, \quad V_C(t) = Q(t)/C, \quad V_L(t) = L\dot{I}(t), \quad I(t) = \dot{Q}(t),$$

gives

$$e(t) = L\ddot{Q}(t) + R\dot{Q}(t) + Q(t)/C.$$

Linear RLC circuit



Kirchhoff's law states that the total voltage must be balanced:

$$e(t) - V_R(t) - V_C(t) - V_L(t) = 0.$$

Combined with

$$V_R(t) = I(t)R, \quad V_C(t) = Q(t)/C, \quad V_L(t) = L\dot{I}(t), \quad I(t) = \dot{Q}(t),$$

gives

$$e(t) = L\ddot{Q}(t) + R\dot{Q}(t) + Q(t)/C.$$

Linear RLC circuit, ctd.

Second order \rightarrow first order by introducing $x_1 = Q$, $x_2 = \dot{Q}$:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -1/LC & -R/L \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1/L \end{bmatrix} e$$

Difficult to observe Q directly, instead current $I = \dot{Q} = x_2$:

$$y = x_2 = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

Typical question of **observability**:

Is it possible to determine the complete state from this observation?

Linear RLC circuit, ctd.

Second order \rightarrow first order by introducing $x_1 = Q$, $x_2 = \dot{Q}$:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -1/LC & -R/L \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1/L \end{bmatrix} e$$

Difficult to observe Q directly, instead current $I = \dot{Q} = x_2$:

$$y = x_2 = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

Typical question of **observability**:

Is it possible to determine the complete state from this observation?

Discrete time control systems

If time evolves in discrete steps $0, 1, 2, 3, \dots$:

$$\begin{aligned}x(k+1) &= f(k, x(k), u(k)), & x(0) &= x_0, \\y(k) &= h(k, x(k), u(k)).\end{aligned}$$

LTI case:

$$\begin{aligned}x(k+1) &= Ax(k) + Bu(k), & x(0) &= x_0, \\y(k) &= Cx(k) + Du(k).\end{aligned}$$

Discrete time control systems

If time evolves in discrete steps $0, 1, 2, 3, \dots$:

$$\begin{aligned}x(k+1) &= f(k, x(k), u(k)), & x(0) &= x_0, \\y(k) &= h(k, x(k), u(k)).\end{aligned}$$

LTI case:

$$\begin{aligned}x(k+1) &= Ax(k) + Bu(k), & x(0) &= x_0, \\y(k) &= Cx(k) + Du(k).\end{aligned}$$

Predator-prey model

$x_1(k)$ = number of rabbits

$x_2(k)$ = number of foxes

a = growth rate of rabbits in absence of predation

b = death rate per encounter of rabbits due to predation

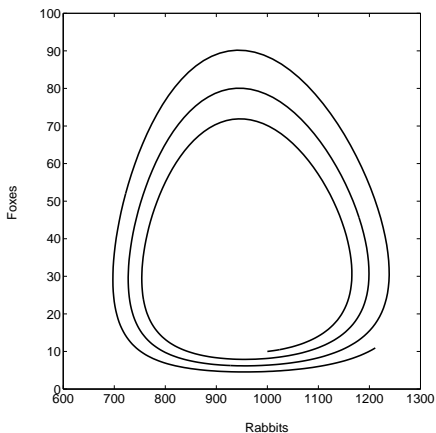
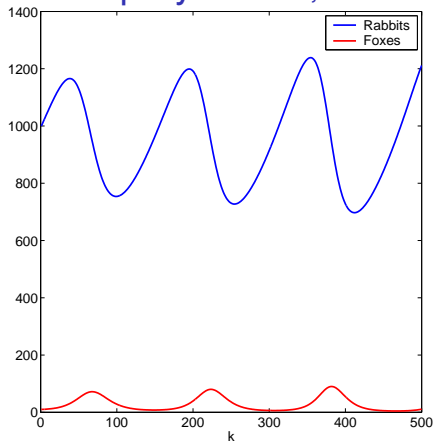
c = death rate of foxes in absence of food

e = efficiency of turning predated rabbits into foxes.

$$x_1(k+1) = x_1(k) + a x_1(k) - b x_1(k)x_2(k)$$

$$x_2(k+1) = x_2(k) + e b x_1(k)x_2(k) - c x_2(k)$$

Predator-prey model, ctd.



Unstable situation! Typical question of **stabilizability**:

Is it possible to stabilize the system by applying control?

System analysis

Introduce concepts of

- ▶ frequency domain approach
- ▶ time domain/state space approach
- ▶ stability
- ▶ controllability
- ▶ observability

In the following, we focus on:

$$\begin{aligned}\dot{x}(t) &= Ax(t) + Bu(t), & x(0) &= x_0, \\ y(t) &= Cx(t) + Du(t).\end{aligned}$$

Laplace transform

For real-valued function $f(t)$, **Laplace transform** is defined as

$$\hat{f}(s) = (\mathcal{L}f)(s) = \int_0^{\infty} f(t) e^{-st} dt.$$

Some properties:

- ▶ $\mathcal{L}(\alpha f + \beta g) = \alpha(\mathcal{L}f) + \beta(\mathcal{L}g)$;
- ▶ $(\mathcal{L}\dot{f})(s) = s(\mathcal{L}f)(s) - f(0)$;
- ▶ $(\mathcal{L} \int f(\tau) d\tau)(s) = \frac{1}{s}(\mathcal{L}f)(s)$.

Some special functions:

- ▶ $\hat{\delta}(s) \equiv 1$, where δ is the Dirac impulse at 0;
- ▶ $f(t) = e^{-\lambda t} \rightsquigarrow \hat{f}(s) = 1/(s + \lambda)$;
- ▶ $f(t) = \sin(bt) \rightsquigarrow \hat{f}(s) = b/(s^2 + b^2)$;
- ▶ $f(t) = \cos(bt) \rightsquigarrow \hat{f}(s) = s/(s^2 + b^2)$.

Laplace transform

For real-valued function $f(t)$, **Laplace transform** is defined as

$$\hat{f}(s) = (\mathcal{L}f)(s) = \int_0^{\infty} f(t) e^{-st} dt.$$

Some properties:

- ▶ $\mathcal{L}(\alpha f + \beta g) = \alpha(\mathcal{L}f) + \beta(\mathcal{L}g)$;
- ▶ $(\mathcal{L}\dot{f})(s) = s(\mathcal{L}f)(s) - f(0)$;
- ▶ $(\mathcal{L} \int f(\tau) d\tau)(s) = \frac{1}{s}(\mathcal{L}f)(s)$.

Some special functions:

- ▶ $\hat{\delta}(s) \equiv 1$, where δ is the Dirac impulse at 0;
- ▶ $f(t) = e^{-\lambda t} \rightsquigarrow \hat{f}(s) = 1/(s + \lambda)$;
- ▶ $f(t) = \sin(bt) \rightsquigarrow \hat{f}(s) = b/(s^2 + b^2)$;
- ▶ $f(t) = \cos(bt) \rightsquigarrow \hat{f}(s) = s/(s^2 + b^2)$.

Laplace transform

For real-valued function $f(t)$, **Laplace transform** is defined as

$$\hat{f}(s) = (\mathcal{L}f)(s) = \int_0^{\infty} f(t) e^{-st} dt.$$

Some properties:

- ▶ $\mathcal{L}(\alpha f + \beta g) = \alpha(\mathcal{L}f) + \beta(\mathcal{L}g)$;
- ▶ $(\mathcal{L}\dot{f})(s) = s(\mathcal{L}f)(s) - f(0)$;
- ▶ $(\mathcal{L} \int f(\tau) d\tau)(s) = \frac{1}{s}(\mathcal{L}f)(s)$.

Some special functions:

- ▶ $\hat{\delta}(s) \equiv 1$, where δ is the Dirac impulse at 0;
- ▶ $f(t) = e^{-\lambda t} \rightsquigarrow \hat{f}(s) = 1/(s + \lambda)$;
 $f(t) = \sin(bt) \rightsquigarrow \hat{f}(s) = b/(s^2 + b^2)$;
 $f(t) = \cos(bt) \rightsquigarrow \hat{f}(s) = s/(s^2 + b^2)$.

Transfer matrices

Applying Laplace transform to $\dot{x}(t) = Ax(t) + Bu(t)$ yields

$$s\hat{x}(s) = A\hat{x}(s) + B\hat{u}(s) \Rightarrow \hat{x}(s) = (sI - A)^{-1}B\hat{u}(s).$$

Inserting into $\hat{y}(s) = C\hat{x}(s) + D\hat{u}(s)$ gives

$$\hat{y}(s) = (C(sI - A)^{-1}B + D)\hat{u}(s).$$

The $p \times m$ matrix $G(s) = C(sI - A)^{-1}B + D$ is rational and called the **transfer function/matrix**.

Eigenvalues of A are **poles**. Zeros of $G(\cdot)$ are **system zeros** (SISO).

For the linear RCL circuit:

$$G(s) = \frac{1/L}{s^2 + R/L \cdot s + 1/LC}$$

Transfer matrices

Applying Laplace transform to $\dot{x}(t) = Ax(t) + Bu(t)$ yields

$$s\hat{x}(s) = A\hat{x}(s) + B\hat{u}(s) \Rightarrow \hat{x}(s) = (sI - A)^{-1}B\hat{u}(s).$$

Inserting into $\hat{y}(s) = C\hat{x}(s) + D\hat{u}(s)$ gives

$$\hat{y}(s) = (C(sI - A)^{-1}B + D)\hat{u}(s).$$

The $p \times m$ matrix $G(s) = C(sI - A)^{-1}B + D$ is rational and called the **transfer function/matrix**.

Eigenvalues of A are **poles**. Zeros of $G(\cdot)$ are **system zeros** (SISO).

For the linear RCL circuit:

$$G(s) = \frac{1/L}{s^2 + R/L \cdot s + 1/LC}$$

Transfer matrices

Applying Laplace transform to $\dot{x}(t) = Ax(t) + Bu(t)$ yields

$$s\hat{x}(s) = A\hat{x}(s) + B\hat{u}(s) \Rightarrow \hat{x}(s) = (sI - A)^{-1}B\hat{u}(s).$$

Inserting into $\hat{y}(s) = C\hat{x}(s) + D\hat{u}(s)$ gives

$$\hat{y}(s) = (C(sI - A)^{-1}B + D)\hat{u}(s).$$

The $p \times m$ matrix $G(s) = C(sI - A)^{-1}B + D$ is rational and called the **transfer function/matrix**.

Eigenvalues of A are **poles**. Zeros of $G(\cdot)$ are **system zeros** (SISO).

For the linear RCL circuit:

$$G(s) = \frac{1/L}{s^2 + R/L \cdot s + 1/LC}$$

Hardy spaces and H^p norms

Appropriate spaces for “Laplace transformed” functions.

Definition

$H^p(\mathbb{C}^m)$ is the space of all analytic $u : \mathbb{C}^+ \rightarrow \mathbb{C}^m$ with $\|u(\cdot)\|_{H^p} < \infty$ for

$$\|u(\cdot)\|_{H^p} = \begin{cases} \sup_{\beta > 0} \left(\int_{-\infty}^{\infty} \|u(\beta + i\omega)\|^p d\omega \right)^{1/p} & \text{if } 1 \leq p < \infty, \\ \sup_{s \in \mathbb{C}^+} \|u(s)\| & \text{if } p = \infty. \end{cases}$$

Important property:

Define boundary function $u^0(i\omega) = \lim_{\beta \downarrow 0} u(\beta + i\omega)$. Then $u^0 \in L^p(i\mathbb{R}, \mathbb{C}^m)$ and $\|u\|_{H^p} = \|u^0\|_{L^p}$. In particular,

$$\|u\|_{H^\infty} = \sup_{\omega \in \mathbb{R}} \|u(i\omega)\|.$$

Hardy spaces and H^p norms

Appropriate spaces for “Laplace transformed” functions.

Definition

$H^p(\mathbb{C}^m)$ is the space of all analytic $u : \mathbb{C}^+ \rightarrow \mathbb{C}^m$ with $\|u(\cdot)\|_{H^p} < \infty$ for

$$\|u(\cdot)\|_{H^p} = \begin{cases} \sup_{\beta > 0} \left(\int_{-\infty}^{\infty} \|u(\beta + i\omega)\|^p d\omega \right)^{1/p} & \text{if } 1 \leq p < \infty, \\ \sup_{s \in \mathbb{C}^+} \|u(s)\| & \text{if } p = \infty. \end{cases}$$

Important property:

Define boundary function $u^0(i\omega) = \lim_{\beta \downarrow 0} u(\beta + i\omega)$. Then $u^0 \in L^p(i\mathbb{R}, \mathbb{C}^m)$ and $\|u\|_{H^p} = \|u^0\|_{L^p}$. In particular,

$$\|u\|_{H^\infty} = \sup_{\omega \in \mathbb{R}} \|u(i\omega)\|.$$

H^p norms, ctd.

Plancherel-like equality (important for transferring results between time and frequency domain):

The normalized Laplace transform $\sqrt{2\pi}\mathcal{L}$ is a linear *isometry* from $L^2((0, \infty); \mathbb{C}^m)$ onto $H^2(\mathbb{C}^m)$.

H^∞ is often the space easier to work with. Transfer function $G(s) \in \mathbb{C}^{p \times m}$:

$$\|G\|_{H^\infty} = \sup_{\omega \in \mathbb{R}} \|G(i\omega)\|_2.$$

(Computational methods closely related to stability radius.)

H^p norms, ctd.

Plancherel-like equality (important for transferring results between time and frequency domain):

The normalized Laplace transform $\sqrt{2\pi}\mathcal{L}$ is a linear *isometry* from $L^2((0, \infty); \mathbb{C}^m)$ onto $H^2(\mathbb{C}^m)$.

H^∞ is often the space easier to work with. Transfer function $G(s) \in \mathbb{C}^{p \times m}$:

$$\|G\|_{H^\infty} = \sup_{\omega \in \mathbb{R}} \|G(i\omega)\|_2.$$

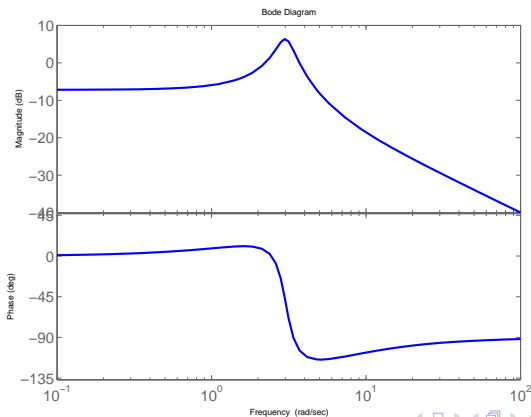
(Computational methods closely related to stability radius.)

Bode plot

SISO case: For input $u(t) = u_0 \sin(\omega t)$, \approx output for large t

$$y^{SS}(t) = u_0 |G(j\omega)| \sin(\omega t + \arg(G(j\omega))).$$

Example: $g(s) = \frac{(s-1)(s+4)}{(s-1)(s+0.4-3i)(s+0.4+3i)}$.



Don't use transfer functions for computations!

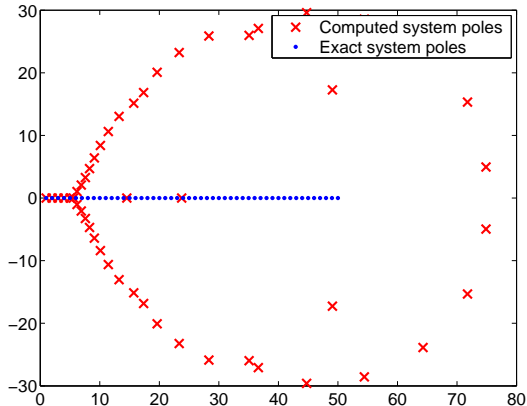
MATLAB script:

```
n = 50;  
A = diag(1:n); B = ones(n,1); C = ones(1,n); D = 0;  
sys = ss(A,B,C,D); tra = tf(sys);  
p = eig(tra);  
plot(p);
```

Don't use transfer functions for computations!

MATLAB script:

```
n = 50;
A = diag(1:n); B = ones(n,1); C = ones(1,n); D = 0;
sys = ss(A,B,C,D); tra = tf(sys);
p = eig(tra);
plot(p);
```



The exponential matrix function

Important tool for developing state space representations:

$$e^A := I + \frac{A}{1!} + \frac{A^2}{2!} + \frac{A^3}{3!} + \dots$$

Some useful properties:

- ▶ $e^{A(t+s)} = e^{At} e^{As}$, $e^{-At} = [e^{At}]^{-1}$, $\frac{\partial}{\partial t} e^{At} = A e^{At}$;
- ▶ $e^{P^{-1}AP} = P^{-1} e^A P$;
- ▶ if λ is an eigenvalue of A then e^λ is an eigenvalue of e^A ;
- ▶ **But:** $e^{A+B} \neq e^A e^B$ (unless A and B commute).

The exponential matrix function

Important tool for developing state space representations:

$$e^A := I + \frac{A}{1!} + \frac{A^2}{2!} + \frac{A^3}{3!} + \dots$$

Some useful properties:

- ▶ $e^{A(t+s)} = e^{At} e^{As}$, $e^{-At} = [e^{At}]^{-1}$, $\frac{\partial}{\partial t} e^{At} = A e^{At}$;
- ▶ $e^{P^{-1}AP} = P^{-1} e^A P$;
- ▶ if λ is an eigenvalue of A then e^λ is an eigenvalue of e^A ;
- ▶ **But:** $e^{A+B} \neq e^A e^B$ (unless A and B commute).

Solution of state equation

The state equation

$$\dot{x}(t) = Ax(t) + Bu(t), \quad x(0) = x_0,$$

is a linear ODE and thus *always* uniquely solvable. Solution:

$$x(t) = e^{At} x_0 + \int_0^t e^{A(t-\tau)} Bu(\tau) d\tau.$$

Inserting into output equation $y(t) = Cx(t) + Du(t)$ yields

$$y(t) = C e^{At} x_0 + \int_0^t C e^{A(t-\tau)} Bu(\tau) d\tau + Du(t).$$

\rightsquigarrow *input-output map* $\mathcal{G} : u(\cdot) \rightarrow y(\cdot)$.

Solution of state equation

The state equation

$$\dot{x}(t) = Ax(t) + Bu(t), \quad x(0) = x_0,$$

is a linear ODE and thus *always* uniquely solvable. Solution:

$$x(t) = e^{At} x_0 + \int_0^t e^{A(t-\tau)} Bu(\tau) d\tau.$$

Inserting into output equation $y(t) = Cx(t) + Du(t)$ yields

$$y(t) = C e^{At} x_0 + \int_0^t C e^{A(t-\tau)} Bu(\tau) d\tau + Du(t).$$

\rightsquigarrow *input-output map* $\mathcal{G} : u(\cdot) \rightarrow y(\cdot)$.

Stability

1. The system $\dot{x}(t) = Ax(t)$ is called **asymptotically stable** if for each x_0 the solution $x(t)$ satisfies $\lim_{t \rightarrow \infty} x(t) = 0$.
2. If for each x_0 there exists a constant $C > 0$ so that $\|x(t)\| \leq C$ for all $t > 0$ then the system is called **stable**.
3. In any other case, the system is called **unstable**.

For $\dot{x}(t) = Ax(t) + Bu(t)$, let $x(t)$ be solution corresponding to initial state x_0 and let $\hat{x}(t)$ be solution corresponding to perturbed \hat{x}_0 . Then the error between both trajectories $e(t) = \hat{x}(t) - x(t)$ satisfies $\dot{e}(t) = Ae(t)$.

Stability

1. The system $\dot{x}(t) = Ax(t)$ is called **asymptotically stable** if for each x_0 the solution $x(t)$ satisfies $\lim_{t \rightarrow \infty} x(t) = 0$.
2. If for each x_0 there exists a constant $C > 0$ so that $\|x(t)\| \leq C$ for all $t > 0$ then the system is called **stable**.
3. In any other case, the system is called **unstable**.

For $\dot{x}(t) = Ax(t) + Bu(t)$, let $x(t)$ be solution corresponding to initial state x_0 and let $\hat{x}(t)$ be solution corresponding to perturbed \hat{x}_0 . Then the error between both trajectories $e(t) = \hat{x}(t) - x(t)$ satisfies $\dot{e}(t) = Ae(t)$.

Eigenvalue criterion for stability

The system $\dot{x}(t) = Ax(t)$ is

1. asymptotically stable $\Leftrightarrow \lambda(A) \subset \mathbb{C}^-$;
2. stable $\Leftrightarrow \lambda(A) \subset \mathbb{C}^- \cup i\mathbb{R}$ and each purely imaginary eigenvalue is semi-simple.

Examples:

$$A = \begin{bmatrix} 0 & \theta \\ -\theta & 0 \end{bmatrix} \Rightarrow e^{At} = \begin{bmatrix} \cos(\theta t) & \sin(\theta t) \\ -\sin(\theta t) & \cos(\theta t) \end{bmatrix}$$

$$A = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \Rightarrow e^{At} = \begin{bmatrix} 1 & t \\ 0 & 1 \end{bmatrix}$$

Eigenvalue criterion for stability

The system $\dot{x}(t) = Ax(t)$ is

1. asymptotically stable $\Leftrightarrow \lambda(A) \subset \mathbb{C}^-$;
2. stable $\Leftrightarrow \lambda(A) \subset \mathbb{C}^- \cup i\mathbb{R}$ and each purely imaginary eigenvalue is semi-simple.

Examples:

$$A = \begin{bmatrix} 0 & \theta \\ -\theta & 0 \end{bmatrix} \Rightarrow e^{At} = \begin{bmatrix} \cos(\theta t) & \sin(\theta t) \\ -\sin(\theta t) & \cos(\theta t) \end{bmatrix}$$

$$A = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \Rightarrow e^{At} = \begin{bmatrix} 1 & t \\ 0 & 1 \end{bmatrix}$$

Lyapunov's direct method

Consider time-invariant system with equilibrium point \bar{x} :

$$\dot{x}(t) = f(x(t)), \quad f(\bar{x}) = 0.$$

Let D be an open neighborhood of \hat{x} . Differentiable $V : D \rightarrow \mathbb{R}$ is called **Lyapunov function** if

$$V(\bar{x}) = 0, \quad V(x) > 0, \quad x \in D \setminus \{\bar{x}\}, \quad \dot{V}(x) \leq 0, \quad x \in D,$$

where $\dot{V}(x) = \langle \text{grad} V(x), f(x) \rangle$.

Existence of Lyapunov function $\Rightarrow \bar{x}$ is **stable**.

V is a **strict Lyapunov function** if additionally $\dot{V}(x) < 0, x \in D \setminus \{\bar{x}\}$.

Existence of strict Lyapunov function $\Rightarrow \bar{x}$ is **asymptotically stable**.

Lyapunov's direct method

Consider time-invariant system with equilibrium point \bar{x} :

$$\dot{x}(t) = f(x(t)), \quad f(\bar{x}) = 0.$$

Let D be an open neighborhood of \hat{x} . Differentiable $V : D \rightarrow \mathbb{R}$ is called **Lyapunov function** if

$$V(\bar{x}) = 0, \quad V(x) > 0, \quad x \in D \setminus \{\bar{x}\}, \quad \dot{V}(x) \leq 0, \quad x \in D,$$

where $\dot{V}(x) = \langle \text{grad} V(x), f(x) \rangle$.

Existence of Lyapunov function $\Rightarrow \bar{x}$ is **stable**.

V is a **strict Lyapunov function** if additionally $\dot{V}(x) < 0, x \in D \setminus \{\bar{x}\}$.

Existence of strict Lyapunov function $\Rightarrow \bar{x}$ is **asymptotically stable**.

Lyapunov function for LTI

For $\dot{x}(t) = Ax(t)$ we take $V(x) = x^T Px$ for some symmetric $P \in \mathbb{R}^{n \times n}$.

$$\dot{V}(x) = x^T A^T Px + x^T PAx = x^T (A^T P + PA)x.$$

For a strict Lyapunov function P and $Q = -(A^T P + PA)$ must be positive definite.

Theorem

$\dot{x}(t) = Ax(t)$ is asymptotically stable \Leftrightarrow the Lyapunov matrix equation

$$A^T P + PA = -Q$$

has a s.p.d. solution P for any s.p.d. Q .

Idea of proof: $P = \int_0^\infty e^{A^T t} Q e^{At} dt$.

Lyapunov function for LTI

For $\dot{x}(t) = Ax(t)$ we take $V(x) = x^T Px$ for some symmetric $P \in \mathbb{R}^{n \times n}$.

$$\dot{V}(x) = x^T A^T Px + x^T PAx = x^T (A^T P + PA)x.$$

For a strict Lyapunov function P and $Q = -(A^T P + PA)$ must be positive definite.

Theorem

$\dot{x}(t) = Ax(t)$ is asymptotically stable \Leftrightarrow the Lyapunov matrix equation

$$A^T P + PA = -Q$$

has a s.p.d. solution P for any s.p.d. Q .

Idea of proof: $P = \int_0^\infty e^{A^T t} Q e^{At} dt$.

Definition of controllability

The system $\dot{x}(t) = Ax(t) + Bu(t)$ is called **controllable** if there is a (continuous) control law $u(t)$ that transfers any initial state x_0 to any x_f at some time $t_f > 0$.

Remarks:

- ▶ State depends linearly on $x_0 \rightsquigarrow$ w.l.o.g. $x_f = 0$.
- ▶ Time-invariance of system $\rightsquigarrow x_0$ can be transferred in any time $t_f > 0$.

Definition of controllability

The system $\dot{x}(t) = Ax(t) + Bu(t)$ is called **controllable** if there is a (continuous) control law $u(t)$ that transfers any initial state x_0 to any x_f at some time $t_f > 0$.

Remarks:

- ▶ State depends linearly on $x_0 \rightsquigarrow$ w.l.o.g. $x_f = 0$.
- ▶ Time-invariance of system $\rightsquigarrow x_0$ can be transferred in any time $t_f > 0$.

Kalman test

(A, B) is controllable if and only if

$$\text{rank}[B, AB, A^2B, \dots, A^{n-1}B] = n.$$

Note: Proof facilitates equivalent condition

$$P(t_f) = \int_0^{t_f} e^{-A\tau} BB^T e^{-A^T\tau} d\tau$$

is p.d. for all/some t_f . $t_f \rightarrow \infty$: Controllability Gramian

$$P = \int_0^{\infty} e^{-A\tau} BB^T e^{-A^T\tau} d\tau$$

is unique solution of

$$AP + PA^T = -BB^T$$

(provided A is stable).

Kalman test

(A, B) is controllable if and only if

$$\text{rank}[B, AB, A^2B, \dots, A^{n-1}B] = n.$$

Note: Proof facilitates equivalent condition

$$P(t_f) = \int_0^{t_f} e^{-A\tau} BB^T e^{-A^T\tau} d\tau$$

is p.d. for all/some t_f . $t_f \rightarrow \infty$: [Controllability Gramian](#)

$$P = \int_0^{\infty} e^{-A\tau} BB^T e^{-A^T\tau} d\tau$$

is unique solution of

$$AP + PA^T = -BB^T$$

(provided A is stable).

Kalman test

(A, B) is controllable if and only if

$$\text{rank}[B, AB, A^2B, \dots, A^{n-1}B] = n.$$

Note: Proof facilitates equivalent condition

$$P(t_f) = \int_0^{t_f} e^{-A\tau} BB^T e^{-A^T\tau} d\tau$$

is p.d. for all/some t_f . $t_f \rightarrow \infty$: **Controllability Gramian**

$$P = \int_0^{\infty} e^{-A\tau} BB^T e^{-A^T\tau} d\tau$$

is unique solution of

$$AP + PA^T = -BB^T$$

(provided A is stable).

Direct application of Kalman test is numerically infeasible!

$$A = \begin{bmatrix} 1 & & & & & \\ 20 & 2 & & & & \\ & \ddots & \ddots & & & \\ & & 20 & 19 & & \\ & & & 20 & 20 & \end{bmatrix}, \quad B = \begin{bmatrix} 20 \\ 0 \\ \vdots \\ \vdots \\ 0 \end{bmatrix}$$

$\mathcal{C} = [B, AB, \dots, A^{19}B]$ (MATLAB command `ctrb`) has singular values

$$2.6 \times 10^{28}, 2.2 \times 10^{28}, \dots, 78.2, 20.$$

\mathcal{C} is **numerically singular**!

Hautus test

Is p an eigenvector of A^T , then $p^T B \neq 0$.

Variations:

1. $\text{rank}([A - \lambda I, B]) = n$ for all $\lambda \in \Lambda(A)$;
2. $\text{rank}([A - \lambda I, B]) = n$ for all $\lambda \in \mathbb{C}$.

For example from above, $[A - 20 \cdot I, B]$ has singular values

36.6, ..., 5.9.

Far from being singular!

Hautus test

Is p an eigenvector of A^T , then $p^T B \neq 0$.

Variations:

1. $\text{rank}([A - \lambda I, B]) = n$ for all $\lambda \in \Lambda(A)$;
2. $\text{rank}([A - \lambda I, B]) = n$ for all $\lambda \in \mathbb{C}$.

For example from above, $[A - 20 \cdot I, B]$ has singular values

36.6, ..., 5.9.

Far from being singular!

Hautus test

Is p an eigenvector of A^T , then $p^T B \neq 0$.

Variations:

1. $\text{rank}([A - \lambda I, B]) = n$ for all $\lambda \in \Lambda(A)$;
2. $\text{rank}([A - \lambda I, B]) = n$ for all $\lambda \in \mathbb{C}$.

For example from above, $[A - 20 \cdot I, B]$ has singular values

36.6, ..., 5.9.

Far from being singular!

System transformations

State space transformation $x \rightarrow T^{-1}x$ for nonsingular T corresponds to system transformation

$$\frac{A \mid B}{C \mid D} \Rightarrow \frac{T^{-1}AT \mid T^{-1}B}{CT \mid D}.$$

Does not change the input-output map. To attain numerical stability, restrict to orthogonal coordinate transformations.

Controllability form for SI systems

Consider single input case ($m = 1$): Let $m = 1$ and Q be an orthogonal matrix which maps B to multiple of e_1 :

$$[A|B] \rightsquigarrow [Q^T A Q | Q^T B] = \left[\begin{array}{cccc|c} \times & \times & \times & \times & \times \\ \times & \times & \times & \times & 0 \\ \times & \times & \times & \times & 0 \\ \times & \times & \times & \times & 0 \end{array} \right].$$

Reduce A further to Hessenberg form.

$$\rightsquigarrow \left[\begin{array}{cccc|c} \times & \times & \times & \times & \times \\ \times & \times & \times & \times & 0 \\ 0 & \times & \times & \times & 0 \\ 0 & \times & \times & \times & 0 \end{array} \right].$$

Controllability form for SI systems

Consider single input case ($m = 1$): Let $m = 1$ and Q be an orthogonal matrix which maps B to multiple of e_1 :

$$[A|B] \rightsquigarrow [Q^T A Q | Q^T B] = \left[\begin{array}{cccc|c} \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \mathbf{0} \\ \times & \times & \times & \times & \mathbf{0} \\ \times & \times & \times & \times & \mathbf{0} \end{array} \right].$$

Reduce A further to Hessenberg form.

$$\rightsquigarrow \left[\begin{array}{cccc|c} \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \mathbf{0} \\ \mathbf{0} & \times & \times & \times & \mathbf{0} \\ \mathbf{0} & \times & \times & \times & \mathbf{0} \end{array} \right].$$

Controllability form for SI systems

$$\rightsquigarrow \left[\begin{array}{cccc|c} \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \mathbf{0} \\ \mathbf{0} & \times & \times & \times & \mathbf{0} \\ \mathbf{0} & \mathbf{0} & \times & \times & \mathbf{0} \end{array} \right].$$

Controllability matrix \mathcal{C} of reduced system is upper triangular with diagonal elements

$$b_1, a_{21}b_1, a_{32}a_{21}b_1, a_{43}a_{32}a_{21}b_1, \dots$$

System is controllable $\Leftrightarrow b_1 \neq 0, a_{21} \neq 0, a_{32} \neq 0, \dots$

Controllability form for SI systems

$$\rightsquigarrow \left[\begin{array}{cccc|c} \times & \times & \times & \times & \times \\ \times & \times & \times & \times & 0 \\ 0 & \times & \times & \times & 0 \\ 0 & \mathbf{0} & \times & \times & 0 \end{array} \right].$$

Controllability matrix \mathcal{C} of reduced system is **upper triangular** with diagonal elements

$$b_1, a_{21} b_1, a_{32} a_{21} b_1, a_{43} a_{32} a_{21} b_1, \dots$$

System is controllable $\Leftrightarrow b_1 \neq 0, a_{21} \neq 0, a_{32} \neq 0, \dots$

Controllability form for SI systems

If subdiagonal element is zero, transformed system can be decomposed into

$$\begin{array}{cc|c} A_{11} & A_{12} & B_1 \\ 0 & A_{22} & 0 \\ \hline C_1 & C_2 & D \end{array}$$

Two subsystems:

$$\begin{array}{c|c} A_{11} & B_1 \\ \hline C_1 & D \end{array} \quad \text{is controllable } (\lambda(A_{11}) \text{ are controllable modes});$$

$$\begin{array}{c|c} A_{22} & 0 \\ \hline C_2 & D \end{array} \quad \text{is uncontrollable } (\lambda(A_{22}) \text{ are uncontrollable modes}).$$

Staircase form for MI systems

There exists an orthogonal matrix Q s.t.

$$[Q^T A Q | Q^T B] = \left[\begin{array}{cccc|c|c} A_{1,1} & A_{1,2} & \cdots & A_{1,k} & A_{1,k+1} & X_1 \\ X_2 & A_{2,2} & & \vdots & \vdots & 0 \\ 0 & \ddots & \ddots & \vdots & \vdots & \vdots \\ & \ddots & X_k & A_{k,k} & A_{k,k+1} & 0 \\ \hline & & 0 & 0 & A_{k+1,k+1} & 0 \end{array} \right],$$

where

A_{ij} are $\rho_i \times \rho_i$ matrices;

X_i are $\rho_i \times \rho_{i-1}$ matrices of full row rank.

Eigenvalues of $A_{k+1,k+1}$ are uncontrollable modes.

Definition of stabilizability

System $\dot{x}(t) = Ax(t) + Bu(t)$ is called **stabilizable** if for any x_0 there is a continuous control $u(t)$ such that $\lim_{t \rightarrow \infty} x(t) = 0$.

Later, we will see that the poles of a controllable system can be placed *anywhere* by $u(t) = -BFx(t)$.

↪ for systems in staircase controllability form

$$\begin{array}{cc|c} A_{11} & A_{12} & B_1 \\ 0 & A_{22} & 0 \\ \hline C_1 & C_2 & D \end{array}$$

By the controllability of (A_{11}, B_1) , the poles of A_{11} can always be placed in the open left half plane by some control. But nothing can be done about the poles of A_{22} .

Stabilizability $\Leftrightarrow \lambda(A_{22}) \in \mathbb{C}^-$.

Definition of stabilizability

System $\dot{x}(t) = Ax(t) + Bu(t)$ is called **stabilizable** if for any x_0 there is a continuous control $u(t)$ such that $\lim_{t \rightarrow \infty} x(t) = 0$.

Later, we will see that the poles of a controllable system can be placed *anywhere* by $u(t) = -BFx(t)$.

↪ for systems in staircase controllability form

$$\begin{array}{cc|c} A_{11} & A_{12} & B_1 \\ 0 & A_{22} & 0 \\ \hline C_1 & C_2 & D \end{array}$$

By the controllability of (A_{11}, B_1) , the poles of A_{11} can always be placed in the open left half plane by some control. But nothing can be done about the poles of A_{22} .

Stabilizability $\Leftrightarrow \lambda(A_{22}) \in \mathbb{C}^-$.

Definition of stabilizability

System $\dot{x}(t) = Ax(t) + Bu(t)$ is called **stabilizable** if for any x_0 there is a continuous control $u(t)$ such that $\lim_{t \rightarrow \infty} x(t) = 0$.

Later, we will see that the poles of a controllable system can be placed *anywhere* by $u(t) = -BFx(t)$.

↪ for systems in staircase controllability form

$$\begin{array}{cc|c} A_{11} & A_{12} & B_1 \\ 0 & A_{22} & 0 \\ \hline C_1 & C_2 & D \end{array}$$

By the controllability of (A_{11}, B_1) , the poles of A_{11} can always be placed in the open left half plane by some control. But nothing can be done about the poles of A_{22} .

Stabilizability $\Leftrightarrow \lambda(A_{22}) \in \mathbb{C}^-$.

Definition of stabilizability

System $\dot{x}(t) = Ax(t) + Bu(t)$ is called **stabilizable** if for any x_0 there is a continuous control $u(t)$ such that $\lim_{t \rightarrow \infty} x(t) = 0$.

Later, we will see that the poles of a controllable system can be placed *anywhere* by $u(t) = -BFx(t)$.

↪ for systems in staircase controllability form

$$\begin{array}{cc|c} A_{11} & A_{12} & B_1 \\ 0 & A_{22} & 0 \\ \hline C_1 & C_2 & D \end{array}$$

By the controllability of (A_{11}, B_1) , the poles of A_{11} can always be placed in the open left half plane by some control. But nothing can be done about the poles of A_{22} .

Stabilizability $\Leftrightarrow \lambda(A_{22}) \in \mathbb{C}^-$.

Definition of observability

System $\dot{x}(t) = Ax(t) + Bu(t)$, $y(t) = Cx(t) + Du(t)$ is called **observable** if there is a t_f such that for arbitrary $u(t)$ we can determine $x(0)$ from knowledge of $u(t)$ and $y(t)$ in $[0, t_f]$.

Remark:

- ▶ For LTI systems, it is sufficient to show statement for one u .

Let $g(t) = y(t) - \int_0^{t_f} Ce^{A(t-\tau)} Bu(\tau) d\tau - Du(t)$, then

$$\int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau x_0 = \int_0^{t_f} e^{A^T \tau} C^T g(\tau) d\tau$$

↪ Observability equivalent to invertibility of **observability Gramian**

$$Q(t_f) = \int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau.$$

Definition of observability

System $\dot{x}(t) = Ax(t) + Bu(t)$, $y(t) = Cx(t) + Du(t)$ is called **observable** if there is a t_f such that for arbitrary $u(t)$ we can determine $x(0)$ from knowledge of $u(t)$ and $y(t)$ in $[0, t_f]$.

Remark:

- ▶ For LTI systems, it is sufficient to show statement for one u .

Let $g(t) = y(t) - \int_0^{t_f} Ce^{A(t-\tau)} Bu(\tau) d\tau - Du(t)$, then

$$\int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau x_0 = \int_0^{t_f} e^{A^T \tau} C^T g(\tau) d\tau$$

↪ Observability equivalent to invertibility of **observability Gramian**

$$Q(t_f) = \int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau.$$

Definition of observability

System $\dot{x}(t) = Ax(t) + Bu(t)$, $y(t) = Cx(t) + Du(t)$ is called **observable** if there is a t_f such that for arbitrary $u(t)$ we can determine $x(0)$ from knowledge of $u(t)$ and $y(t)$ in $[0, t_f]$.

Remark:

- ▶ For LTI systems, it is sufficient to show statement for one u .

Let $g(t) = y(t) - \int_0^{t_f} Ce^{A(t-\tau)} Bu(\tau) d\tau - Du(t)$, then

$$\int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau x_0 = \int_0^{t_f} e^{A^T \tau} C^T g(\tau) d\tau$$

↪ Observability equivalent to invertibility of **observability Gramian**

$$Q(t_f) = \int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau.$$

Definition of observability

System $\dot{x}(t) = Ax(t) + Bu(t)$, $y(t) = Cx(t) + Du(t)$ is called **observable** if there is a t_f such that for arbitrary $u(t)$ we can determine $x(0)$ from knowledge of $u(t)$ and $y(t)$ in $[0, t_f]$.

Remark:

- ▶ For LTI systems, it is sufficient to show statement for one u .

Let $g(t) = y(t) - \int_0^{t_f} Ce^{A(t-\tau)} Bu(\tau) d\tau - Du(t)$, then

$$\int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau x_0 = \int_0^{t_f} e^{A^T \tau} C^T g(\tau) d\tau$$

↪ Observability equivalent to invertibility of **observability Gramian**

$$Q(t_f) = \int_0^{t_f} e^{A^T \tau} C^T Ce^{A\tau} d\tau.$$

Duality to controllability

Compare both Gramians:

$$Q(t_f) = \int_0^{t_f} e^{A^T \tau} C^T C e^{A \tau} d\tau, \quad P(t_f) = \int_0^{t_f} e^{-A \tau} B B^T e^{-A^T \tau} d\tau$$

Replacing A by $-A^T$ and B by C^T in P yields Q .

Gives equivalent conditions for observability:

1. Observability matrix $\mathcal{O} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}$ has full rank.
2. $\text{rank} \begin{bmatrix} A - \lambda I \\ C \end{bmatrix} = n$ for all $\lambda \in \lambda(C)$.
3. Solution of $A^T Q + QA = -C^T C$ is s.p.d. (provided A is stable).

Duality to controllability

Compare both Gramians:

$$Q(t_f) = \int_0^{t_f} e^{A^T \tau} C^T C e^{A \tau} d\tau, \quad P(t_f) = \int_0^{t_f} e^{-A \tau} B B^T e^{-A^T \tau} d\tau$$

Replacing A by $-A^T$ and B by C^T in P yields Q .

Gives equivalent conditions for observability:

1. Observability matrix $\mathcal{O} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}$ has full rank.

2. $\text{rank} \begin{bmatrix} A - \lambda I \\ C \end{bmatrix} = n$ for all $\lambda \in \lambda(C)$.

3. Solution of $A^T Q + Q A = -C^T C$ is s.p.d. (provided A is stable).

Duality to controllability

Compare both Gramians:

$$Q(t_f) = \int_0^{t_f} e^{A^T \tau} C^T C e^{A \tau} d\tau, \quad P(t_f) = \int_0^{t_f} e^{-A \tau} B B^T e^{-A^T \tau} d\tau$$

Replacing A by $-A^T$ and B by C^T in P yields Q .

Gives equivalent conditions for observability:

1. Observability matrix $\mathcal{O} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}$ has full rank.
2. $\text{rank} \begin{bmatrix} A - \lambda I \\ C \end{bmatrix} = n$ for all $\lambda \in \lambda(C)$.
3. Solution of $A^T Q + QA = -C^T C$ is s.p.d. (provided A is stable).

Observability form for SO systems

By a variation of algorithm for controllability form:

$$\left[\begin{array}{c} Q^T A Q \\ \hline C Q \end{array} \right] = \left[\begin{array}{cccc} \times & \times & \times & \times \\ \times & \times & \times & \times \\ \mathbf{0} & \times & \times & \times \\ \mathbf{0} & \mathbf{0} & \times & \times \\ \hline \mathbf{0} & \mathbf{0} & \mathbf{0} & \times \end{array} \right]$$

If subdiagonal element is zero, decomposition

$$\begin{array}{cc|c} A_{11} & A_{12} & B_1 \\ 0 & A_{22} & B_2 \\ \hline 0 & C_2 & D \end{array}$$

Subsystem $\begin{array}{c|c} A_{11} & B_1 \\ \hline 0 & D \end{array}$ is not observable
 ($\lambda(A_{11})$ unobservable modes).

Observability form for SO systems

By a variation of algorithm for controllability form:

$$\left[\begin{array}{c} Q^T A Q \\ \hline C Q \end{array} \right] = \left[\begin{array}{cccc} \times & \times & \times & \times \\ \times & \times & \times & \times \\ 0 & \times & \times & \times \\ 0 & 0 & \times & \times \\ \hline 0 & 0 & 0 & \times \end{array} \right]$$

If subdiagonal element is zero, decomposition

$$\begin{array}{cc|c} A_{11} & A_{12} & B_1 \\ 0 & A_{22} & B_2 \\ \hline 0 & C_2 & D \end{array}$$

Subsystem $\begin{array}{c|c} A_{11} & B_1 \\ \hline 0 & D \end{array}$ is not observable
 ($\lambda(A_{11})$ **unobservable modes**).

Combining controllability and observability forms

There is an orthogonal matrix Q s.t.

$$\left[\begin{array}{c|c} Q^T A Q & Q^T B \\ \hline C Q & D \end{array} \right] = \left[\begin{array}{ccc|c} A_{c\bar{o}} & \times & \times & B_{c\bar{o}} \\ 0 & A_{co} & \times & B_{co} \\ 0 & 0 & A_{\bar{c}} & 0 \\ \hline 0 & C_{co} & C_{\bar{o}} & D \end{array} \right]$$

where

$A_{c\bar{o}}$ contains unobservable modes;

$A_{\bar{c}}$ contains uncontrollable modes.

Subsystem $\frac{A_{co} \mid B_{co}}{C_{co} \mid D}$ is **controllable and observable**.